### STRUCTURAL DESIGN USING FINITE ELEMENTS

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# INTRODUCTION

- FEA: determining the response of a given structure for a given set of loads and boundary conditions
  - Geometry, material properties, BCs and loads are well defined
- Engineering design: a process of synthesis in which parts are put together to build a structure that will perform a given set of functions satisfactorily
- Analysis is very systematic and can be taught easily; design is an iterative process
- Creative design: creating a new structure or machine that does not exist
- Adaptive design: modifying an existing design (evolutionary process)

# INTRODUCTION - STRUCTURAL DESIGN

- Structural design: a procedure to improve or enhance the performance of a structure by changing its parameters
- Performance: a measurable quantity (constraint and goal)
  - the weight, stiffness or compliance; the fatigue life; noise and vibration levels; safety
- Constraint: As long as the performance satisfies the criterion, its level is not important
  - Ex: the maximum stress should be less than the allowable stress
- Goal: the performance that the engineer wants to improve as much as possible
- Design variables: system parameters that can be changed during the design process
  - Plate thickness, cross-sectional area, shape, etc

# EXAMPLE

- Design the height h of cantilevered beam with  $S_F = 1.5$ -  $E = 2.9 \times 10^4$  ksi, w = 2.25 in.
- 1) Allowable tip deflection  $D_{\text{allowable}} = 2.5$  in. (No need  $S_F$ )
  - FE equation after applying BCs

$$\frac{EI}{L^3} \begin{bmatrix} 12 & -6L \\ -6L & 4L^2 \end{bmatrix} \begin{cases} v_2 \\ \theta_2 \end{cases} = \begin{cases} F \\ 0 \end{cases}$$

- FE solution

$$v_2 = \frac{4FL^3}{Ewh^3}, \quad \theta_2 = \frac{6FL^2}{Ewh^3}$$

$$v_{2} = \frac{4FL^{3}}{Ewh^{3}} = D_{\text{allowable}}$$
$$h = \sqrt[3]{\frac{4FL^{3}}{EwD_{\text{allowable}}}} = 3.66 \text{ in}$$

$$L = 100 \text{ in}$$

$$W$$

$$F = 2,000 \text{ lb}$$

#### EXAMPLE cont.

2) Failure strength = 40 ksi (Need  $S_F$ )

- Supporting moment at the wall

$$\mathcal{C}_{1} = \frac{\mathcal{E}\mathcal{I}}{\mathcal{L}^{3}} \Big[ 6\mathcal{L}\mathcal{V}_{1} + 4\mathcal{L}^{2}\theta_{1} - 6\mathcal{L}\mathcal{V}_{2} + 2\mathcal{L}^{2}\theta_{2} \Big] = -\mathcal{F}\mathcal{L}$$

- Maximum stress at the wall

$$\sigma_{\max} = \frac{M\frac{h}{2}}{I} = \frac{6FL}{wh^2}$$

- Height calculation with the factor of safety

$$\frac{6FL}{wh^2} = \frac{\sigma_F}{S_F} \implies h = \sqrt{\frac{6FLS_F}{w\sigma_F}} = 4.47 \text{ in}$$

#### FSD EXAMPLE - CANTILEVERED BEAM

- w = 2.25, h = 3.5 in. Determine new height using FSD
- Section modulus and max. stress at the initial design

$$S_{old} = \frac{2I}{h} = \frac{wh^2}{6} = \frac{2.25 \times 3.5^2}{6} = 4.594 \text{ in}^3$$
  
$$\sigma_{max} = \frac{M}{S_{old}} = 43.537 \text{ ksi}$$

New section modulus using stress ratio resizing

$$S_{\text{new}} = S_{\text{old}} \frac{\sigma_{\text{max}}}{\sigma_{\text{allowable}}} = 4.594 \times \frac{43.537}{26.667} = 7.5 \text{ in}^3$$

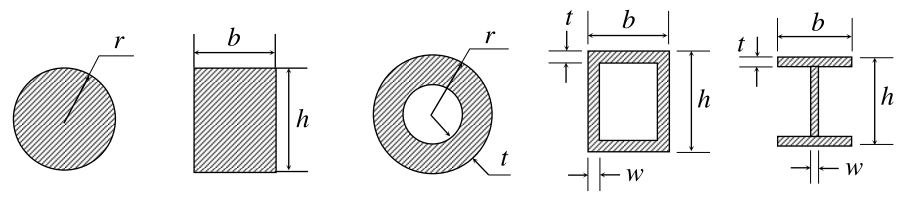
$$S_{\text{new}} = \frac{wh^2}{6} \implies h = \sqrt{\frac{6S_{\text{new}}}{w}} = 4.47 \text{ in}$$

$$L = 100 \text{ in}$$

$$F = 2,000 \text{ lb}$$

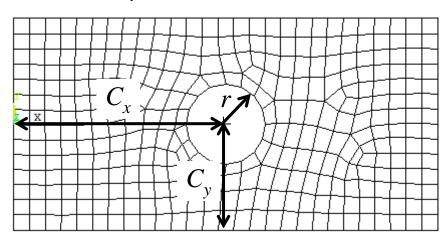
# DESIGN PARAMETERS

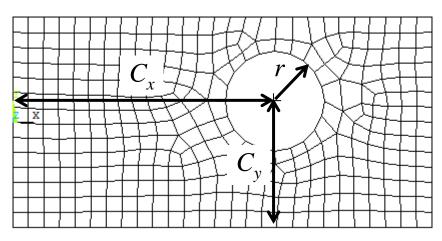
- Selecting design variables easy for beam and truss, but more complicated for plane or 3D solids
- Material property design variable
  - Varying material properties to find the best material
  - Not common, but useful for designing composite materials
- · Sizing design variable
  - Geometric parameters as design (parametric design variable)
  - Appears as a parameter in FEM
  - Thickness of plate/shell, cross-sectional geometry of truss/beam, etc



# DESIGN PARAMETERS cont.

- Shape design variable
  - Related to the structure's geometry, which does not appear explicitly as a parameter
  - Beam cross-section is a geometry, but it appears as a moment of inertia
  - $C_{x}$ ,  $C_{y}$ , and r determine the size and location of the hole
  - Shape design variables change FE mesh
  - Design variables must be limited so that the hole remains inside of the plate





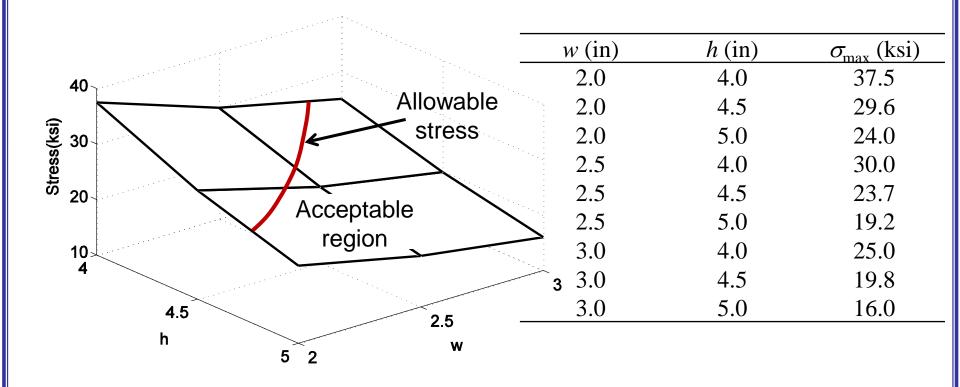
(a) Initial design

(b) Perturbed design

# PARAMETER STUDY - SENSITIVITY ANALYSIS

#### Parameter study

- Effect of a design variable on performance (gradual change of DV)
- Cantilevered beam example:



# SENSITIVITY ANALYSIS

- Parameter study becomes too expensive with many DVs
- Unable to capture rapid change in performance locally
- Design sensitivity analysis computes the rate of performance change with respect to design variables
- Sensitivity analysis calculates gradient of performance for optimization
- Explicit dependence
  - Analytical relationship exists between performance and DVs
  - Weight of circular cross-section beam

$$W(r) = \pi r^2 L$$

- Sensitivity w.r.t. 
$$r: \frac{dW}{dr} = 2\pi rL$$

# SENSITIVITY ANALYSIS cont.

- Implicit dependence
  - Performance depends on DVs through state variable (displacement)
  - Sensitivity of stress:

 $\frac{\mathrm{d}\sigma}{\mathrm{d}r} = \frac{\mathrm{d}\sigma}{\mathrm{d}\mathbf{q}} \cdot \frac{\mathrm{d}\mathbf{q}}{\mathrm{d}r}$ 

Difficult to calculate, time consuming

Easy to calculate from given expression of stress

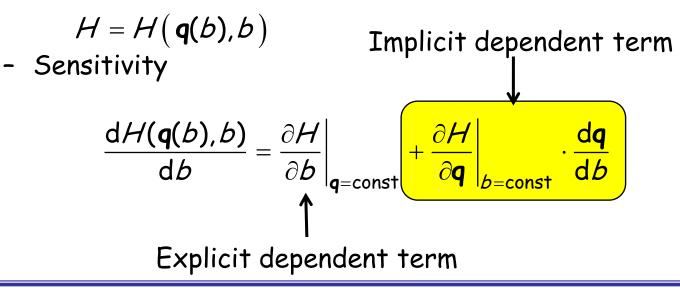
- How to calculate displacement sensitivity?
  - Differentiate finite element equation:  $[K(b)]{Q} = {F(b)}$

$$\begin{bmatrix} \mathbf{K} \end{bmatrix} \left\{ \frac{d\mathbf{Q}}{db} \right\} = \left\{ \frac{d\mathbf{F}}{db} \right\} - \begin{bmatrix} \frac{d\mathbf{K}}{db} \end{bmatrix} \{\mathbf{Q}\}$$

 [dK/db] and {dF/db} can be evaluated using either their analytical expression or numerical differentiation

# SENSITIVITY ANALYSIS cont.

- Sensitivity equation must be solved for each DV
- Sensitivity equation uses the same stiffness matrix with the original finite element analysis
- Consider RHS as a pseudo-force vector
- Similar to finite element analysis with multiple load cases
- Thus, solving sensitivity equation is very inexpensive using factorized stiffness matrix
- General form of performance



# FINITE DIFFERENCE SENSITIVITY

- Easiest way to compute sensitivity information of the performance
- Calculate performance at two different designs
- Forward difference method

$$\frac{\mathrm{d}H}{\mathrm{d}b} \approx \frac{H(b+\Delta b)-H(b)}{\Delta b}$$

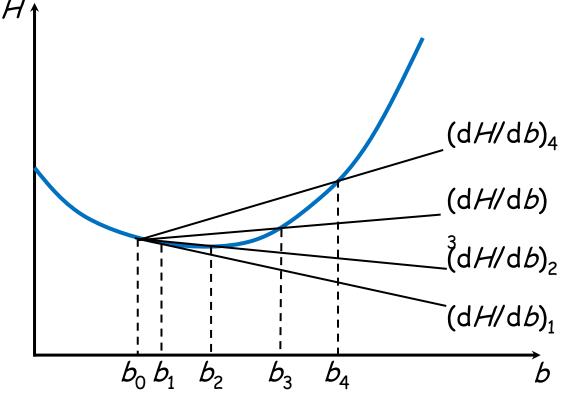
Central difference method

$$\frac{\mathrm{d}H}{\mathrm{d}b}\approx\frac{H(b+\Delta b)-H(b-\Delta b)}{2\Delta b}$$

- Consider FEA as a black-box
- Sensitivity computation cost becomes high for many design variables
  - N+1 analyses for forward FDM
  - 2N+1 analyses for central FDM

# FINITE DIFFERENCE SENSITIVITY cont.

- Accuracy of finite difference sensitivity
  - Accurate results can be expected when  $\Delta b$  approaches zero
  - For nonlinear performances, a large perturbation yields completely inaccurate results
  - Numerical noise becomes dominant for a too-small perturbation size



## EXAMPLE - CANTILEVERED BEAM

- At optimum design (w=2.25 in, h=4.47 in), calculate sensitivity of tip displacement w.r.t. h
- Exact sensitivity:

 $\frac{dv_2}{dh}\Big|_{exact} = -\frac{12FL^3}{Ewh^4} = -\frac{12 \times 2,000 \times 100^3}{2.9 \times 10^7 \times 2.25 \times 4.47^3} = -4.118$ 

- Differentiate [K]
  - $\begin{bmatrix} \frac{d\mathbf{K}}{db} \end{bmatrix} \{\mathbf{Q}\} = \frac{F}{4Lh} \begin{bmatrix} 12 & 6L & -12 & 6L \\ 6L & 4L^2 & -6L & 2L^2 \\ -12 & -6L & 12 & -6L \\ 6L & 2L^2 & -6L & 4L^2 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 4L \\ 6 \end{bmatrix} = \frac{F}{4h} \begin{bmatrix} -12 \\ -12L \\ 12 \\ 0 \end{bmatrix}$
- Pseudo load vector

$$\left\{\frac{\mathrm{d}\mathbf{F}}{\mathrm{d}b}\right\} - \left[\frac{\mathrm{d}\mathbf{K}}{\mathrm{d}b}\right] \left\{\mathbf{Q}\right\} = \frac{F}{4h} \left\{\begin{array}{c}12\\12L\\-12\\0\end{array}\right\}$$

#### EXAMPLE - CANTILEVERED BEAM cont.

Sensitivity equation:

$$\frac{EI}{L^{3}}\begin{bmatrix}
12 & 6L & -12 & 6L \\
6L & 4L^{2} & -6L & 2L^{2} \\
-12 & -6L & 12 & -6L \\
6L & 2L^{2} & -6L & 4L^{2}
\end{bmatrix} \begin{bmatrix}
dv_{1} / db = 0 \\
d\theta_{1} / db = 0 \\
dv_{2} / db \\
d\theta_{2} / db
\end{bmatrix} = \frac{F}{4h} \begin{bmatrix}
12 \\
12L \\
-12 \\
0
\end{bmatrix}$$

- Same way of applying BC  $\frac{EI}{L^{3}} \begin{bmatrix} 12 & -6L \\ -6L & 4L^{2} \end{bmatrix} \begin{bmatrix} dv_{2} / db \\ d\theta_{2} / db \end{bmatrix} = \frac{F}{4h} \begin{bmatrix} -12 \\ 0 \end{bmatrix}$
- Sensitivity of nodal DOFs

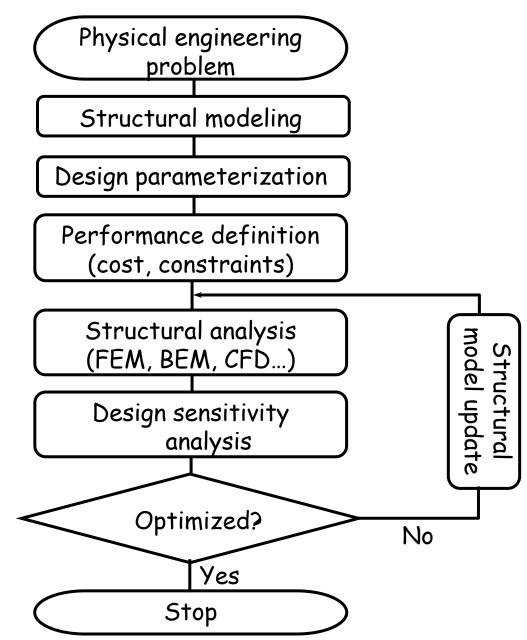
$$\frac{\mathrm{d}v_2}{\mathrm{d}b} = -\frac{12FL^3}{Ewh^4}, \quad \frac{\mathrm{d}\theta_2}{\mathrm{d}b} = -\frac{18FL^2}{Ewh^3}$$

- Same with the exact sensitivity

# STRUCTURAL OPTIMIZATION

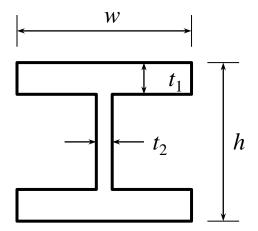
- What Is Design Optimization?
  - To find the best design parameters that meet the design goal and satisfies constraints.
- Design Parameters: Anything the Designer Can Change
  - Thickness of a plate
  - Cross-sectional geometry of a beam or truss
  - Geometric dimensions
- Design Goal: Objective Function
  - Design criterion that will be minimized (or maximized)
  - Mass, Stress, Displacement, Natural Frequency, ETC
- Constraint: Conditions that the system must satisfy
  - Stress, Displacement, ETC
- Note: Design parameters must affect the design goal and constraints.

## OPTIMIZATION FLOW CHART



# THREE-STEP PROBLEM FORMULATION

- 1. Design Parameterization
  - Clear identification
  - Independence of designs



- 2. Objective Function
  - Must be a function of design parameters
  - Minimization (-Maximization)
- 3. Constraint Functions
  - Inequality constraints
  - Equality constraints
  - Equality constraints must be less than the number of design parameters

## STANDARD FORM

#### Standard form of design optimization

 $\begin{array}{ll} \text{minimize} & f(\mathbf{b}) \\ \text{subject to} & g_i(\mathbf{b}) \leq 0, \ i = 1, \cdots, N \\ & h_j(\mathbf{b}) = 0, \ j = 1, \cdots, M \\ & b_l^L \leq b_l \leq b_l^U, \ l = 1, \cdots, K \end{array}$ 

• Feasible set: the set of designs that satisfy constraints

 $\mathcal{S} = \left\{ \mathbf{b} \middle| \mathcal{G}_i(\mathbf{b}) \leq 0, i = 1, \cdots \mathcal{N}, \ h_j(\mathbf{b}) = 0, j = 1, \cdots \mathcal{M} \right\}$